

UNSTEADY ASSESSMENTS AND IMPROVEMENTS FOR THE NATIONAL TRANSONIC FACILITY



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OUTLINE



- **OBJECTIVES**
- **OVERVIEW OF NTF STARBUKS AND FIDO EFFORTS**
- **NTF CENTERLINE PIPE TEST**
- **NTF RAKE TEST**
- **NTF COMMON RESEARCH MODEL TEST**
- **CONCLUDING REMARKS**

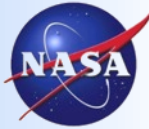
AERIAL VIEW OF NTF



National Transonic Facility
NASA Langley Research Center



WIND TUNNEL IMPROVEMENTS OBJECTIVES FOCUS ON NTF



<u>General Wind Tunnel Requirements</u>	
No. 1	Accuracy and Validation (Repeatability/Data Quality) – Results that can be trusted
No. 2	Productivity – Complete required testing in a timely manner
No. 3	Reliability – Keeps working without interruption

NTF – Recent Efforts

- Subsonic Transonic Applied Refinements By Using Key Strategies (**STARBUKS**)
- Facility Improvements and Data Optimization (**FIDO**)



STARBUKS SUMMARY



Accuracy & Validation

- ✓ Data Acquisition System (Test SLATE)
- ✓ Mach Measurement System
- ✓ Facility Automation System
- ✓ Cooling Coil Trailing Edge Fairings
- ✓ Fixed Fairing Extension
- ✓ Alt. Probes Location (RTD on Cooling Coil)
- ✓ Test Section Visibility
- ✓ Balance Calibrations

Productivity

- ✓ Cryogenic Active Damper
- ✓ Balance Limit Alarm (BLAMS) Upgrade
- ✓ Inlet Guide Vane (IGV) ΔT Mitigation
- ✓ Continuous Pitch

Reliability

- ✓ High Pressure Air Reducing Station
- ✓ Drive Coupling
- ✓ IGV Hydraulic Pipe Repair

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Phase I Testing

✓ Check
Std
Test 214

Check Standard
Model

✓ CRM
Test 215

Common Research
Model

✓ Completed

Phase II Testing

✓ Flow
Calibration
Test 217

CRM Data Flow
Quality
✓ Test 218

See Paryz AIAA 2014-1481



FIDO IMPROVEMENTS ROADMAP



Accuracy & Validation

- ☐ Tunnel configuration selection
- ☐ Free stream Turbulence (Rake – TBD)
- ☒ Mach stability ± 0.0005
- ☒ Conditional sampling (off-line)
- ☒ Validate RTD array on cooling coil

Productivity

- ☐ Mach control methodology
- ☐ 2nd throat actuation
- ☐ Conditional sampling (on-line, real-time)
- ☐ Increase access housing heating
- ☐ Optimized nitrogen injection
- ☒ Continuous sweep

Reliability

- ☐ Liquid nitrogen pump health monitoring
- ☐ Minimize nitrogen system hammering

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Phase I Testing

✓ Check
Std
Test 219

✓ Flow
Survey
Rake
Test 216A

✓ Completed

Phase II Testing

Calibration
Extension
Test 220

CRM
Test 221

Turbulence
Survey Rake
Test 216B

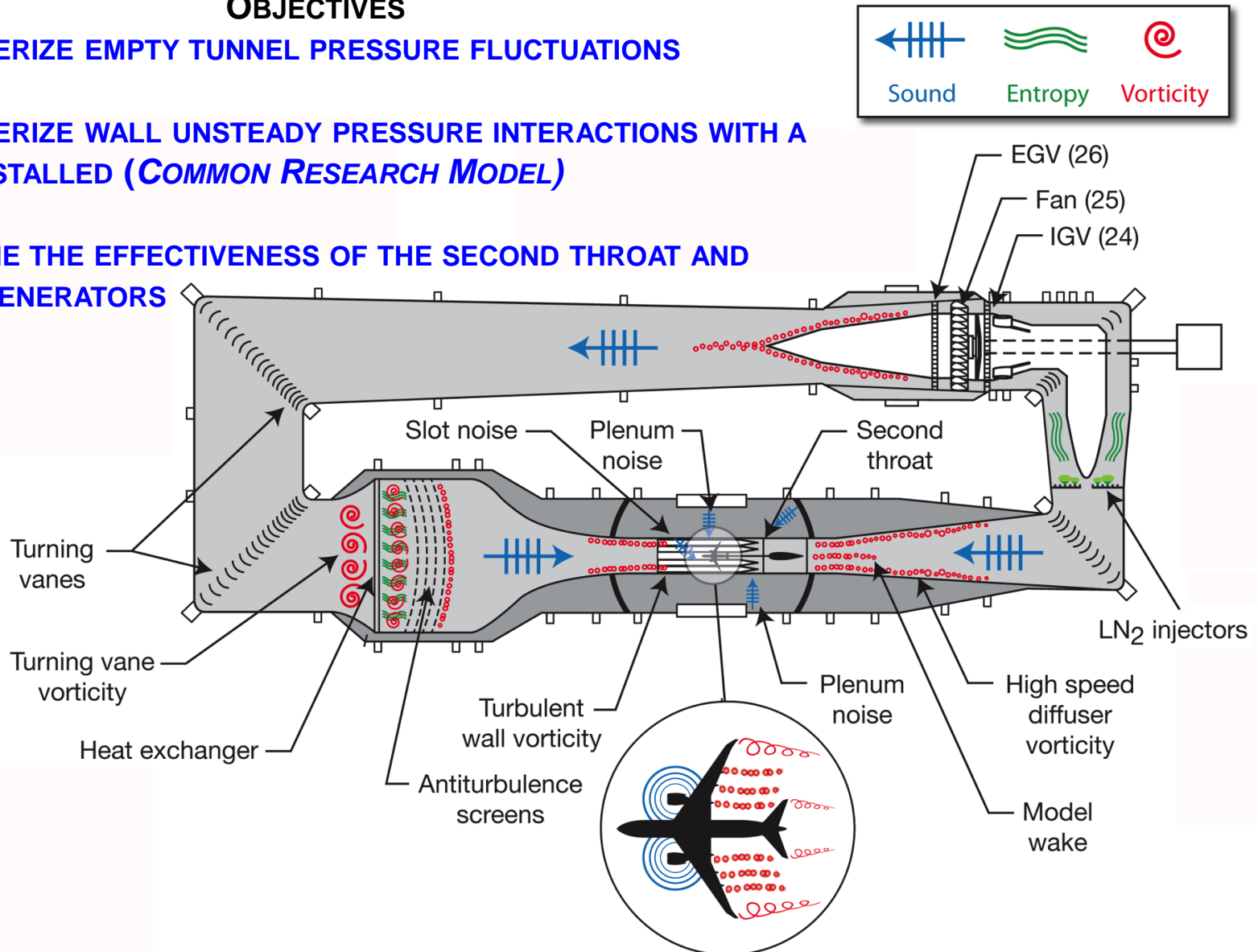


NTF UNSTEADY CHARACTERISTICS THAT CAN INFLUENCE MACH STABILITY AND DATA QUALITY



OBJECTIVES

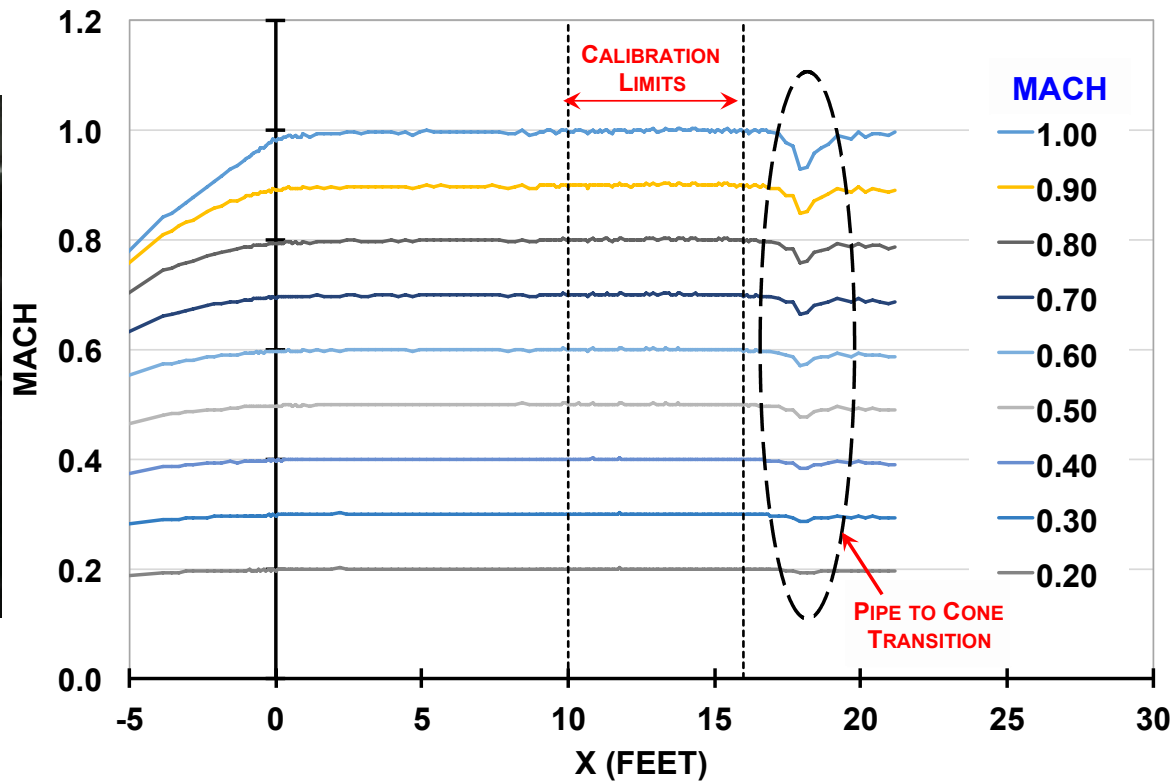
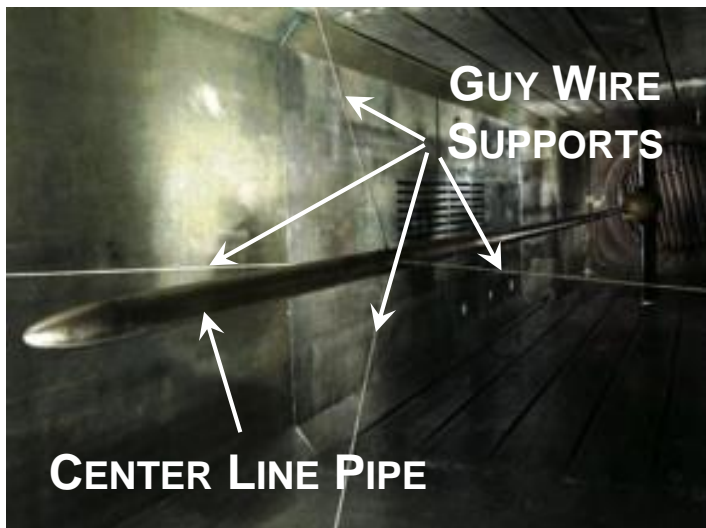
- CHARACTERIZE EMPTY TUNNEL PRESSURE FLUCTUATIONS
- CHARACTERIZE WALL UNSTEADY PRESSURE INTERACTIONS WITH A MODEL INSTALLED (*COMMON RESEARCH MODEL*)
- DETERMINE THE EFFECTIVENESS OF THE SECOND THROAT AND VORTEX GENERATORS



CENTERLINE PIPE TEST



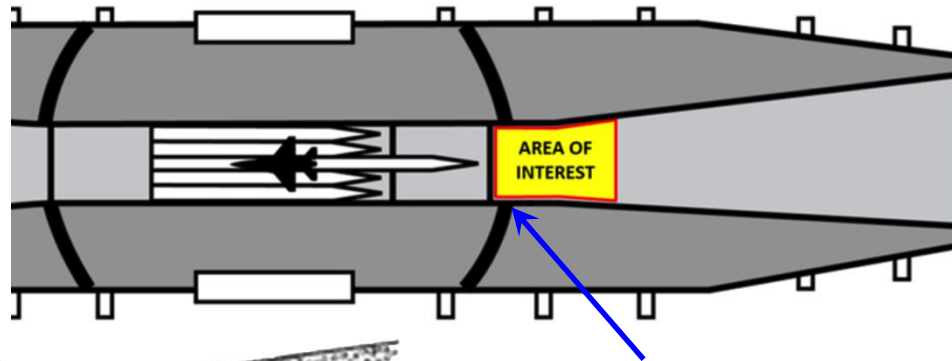
CENTERLINE PIPE DATA TO BE USED FOR TUNNEL CALIBRATION



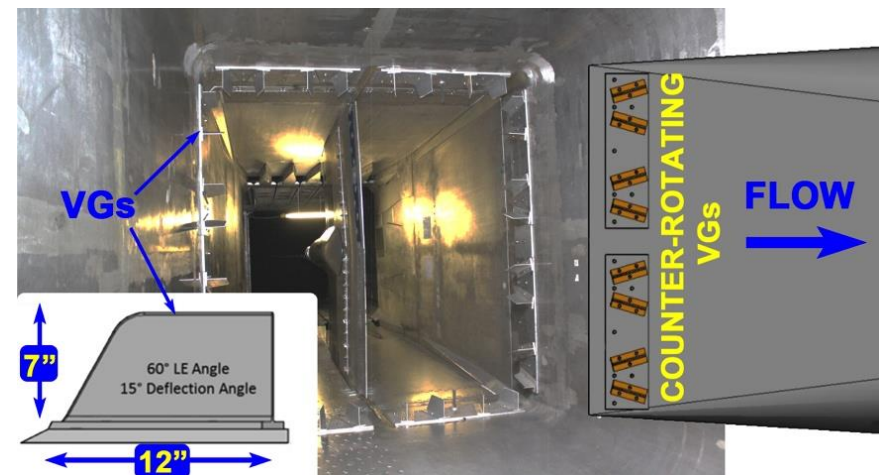
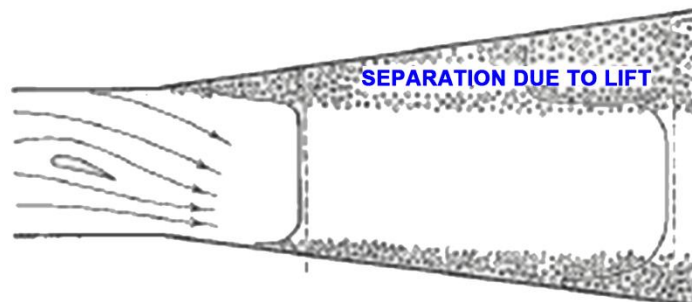
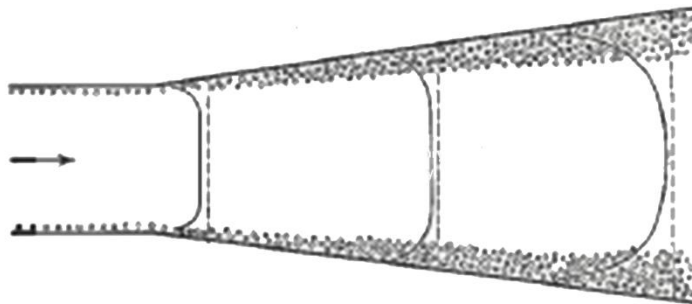
VORTEX GENERATORS



VORTEX GENERATORS ADDED TO MANAGE HIGH SPEED DIFFUSER SEPARATION



VGs LOCATED AT STATION 36'

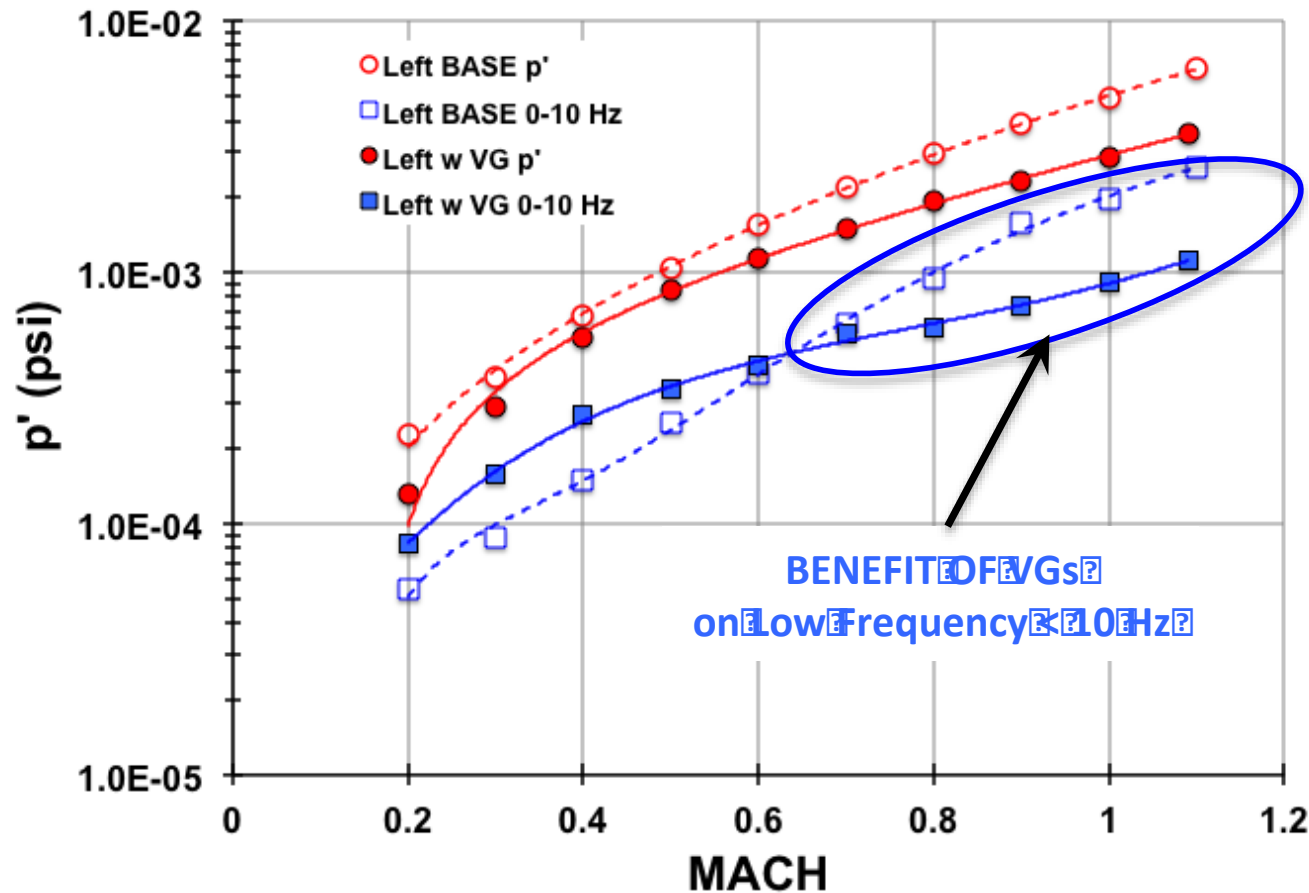


VG SIZE AND SHAPE ARE NOT OPTIMIZED

BENEFITS OF VGs EMPTY TUNNEL



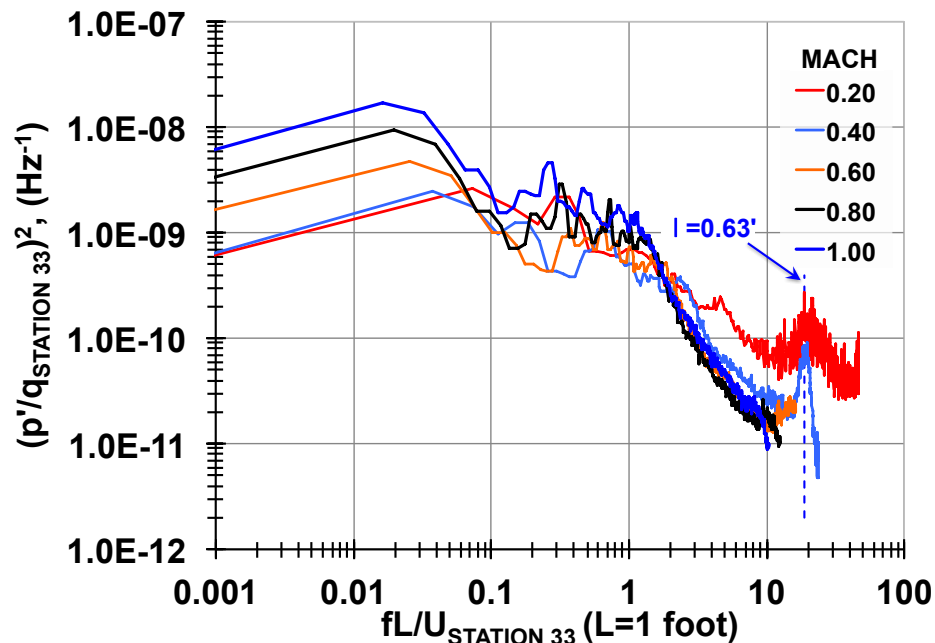
VGs BENEFIT LOW FREQUENCY STABILITY ABOVE MACH ~ 0.7



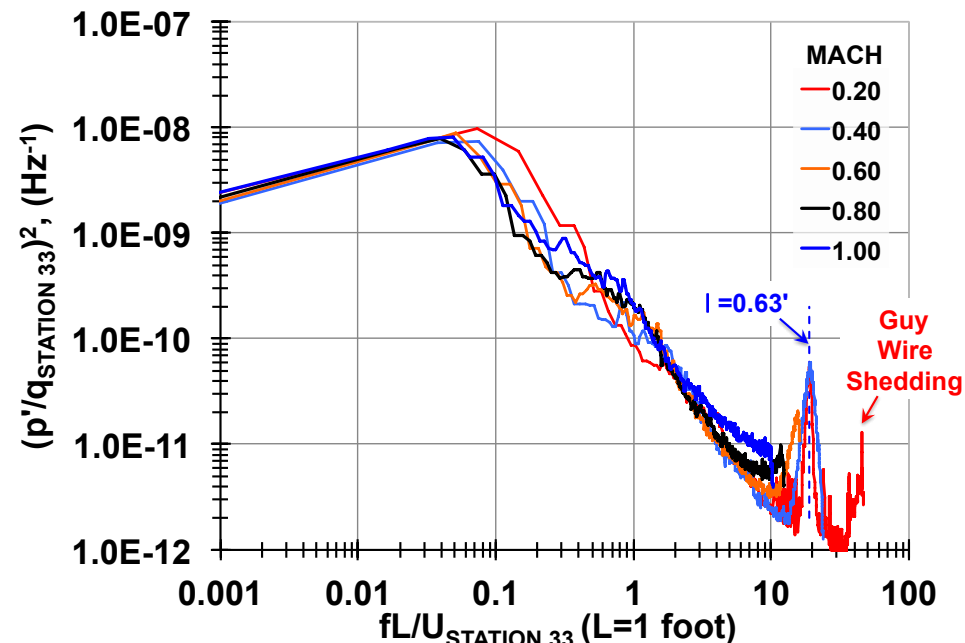
BENEFITS OF VGs (SPECTRA) EMPTY TUNNEL



**VGs SHOW BENEFIT FOR EMPTY TUNNEL
(STABILIZES LOW FREQUENCY AND REDUCES BROADBAND)**

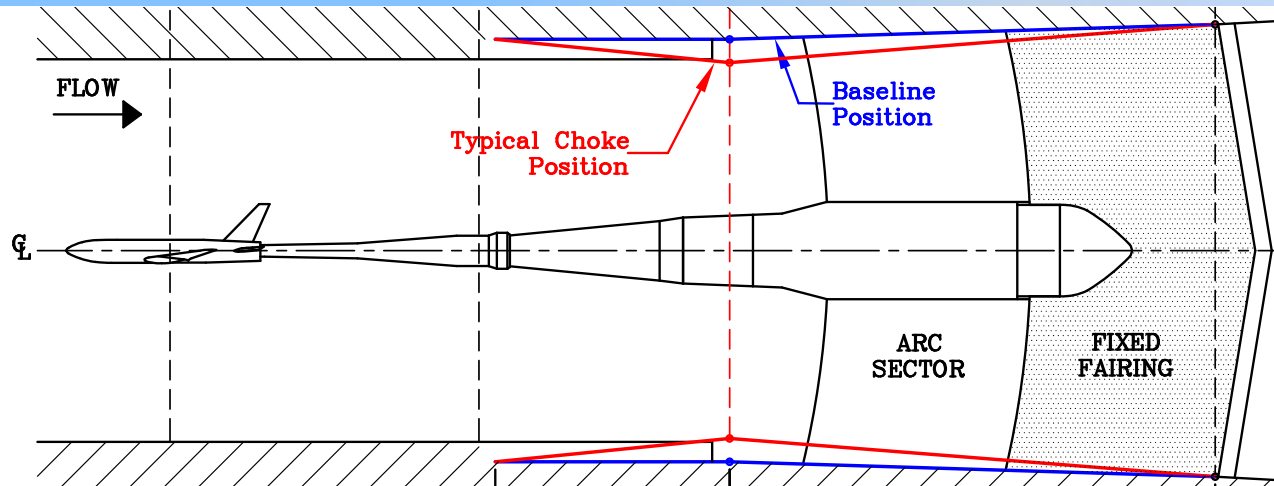


BASELINE



w/ VGs

TEST SECTION MOVABLES (2nd THROAT)

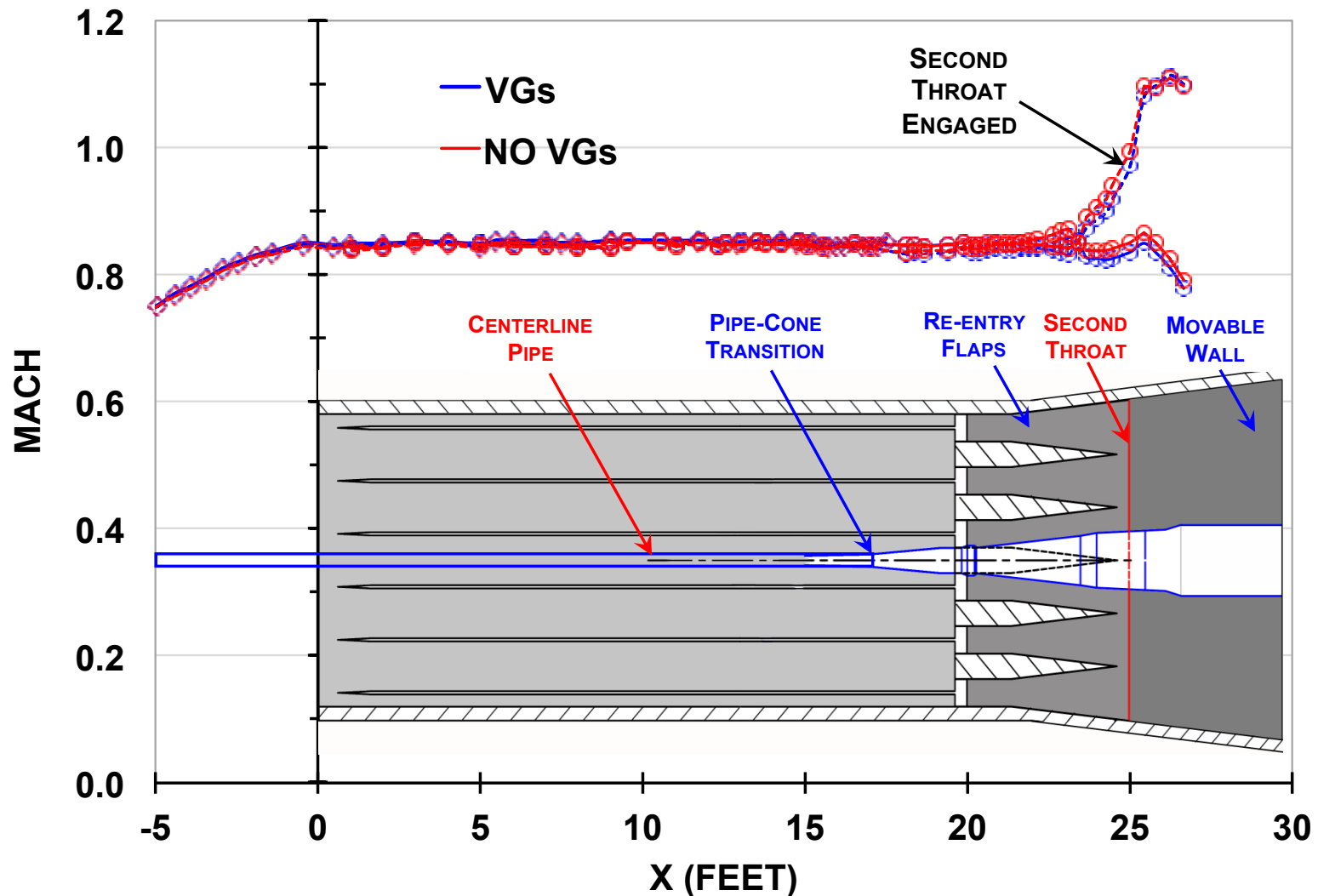


- **Improve Mach stability goals**
 - Target ± 0.0005 Mach number for transonic conditions
 - Determine the influence of Mach stability on performance repeatability (e.g., $C_d \sim 0.0001$)
- **2nd Throat development**
 - Develop a remote positioning system
 - Develop a motorized 2nd throat with a robust instrumentation package to determine wall position
 - Develop a remote wedge system for the fixed faring to minimize support system induced dynamics
- **Planned to be operational in Summer 2015**
- **Requires calibration extension**



Chan AIAA 2015-0622 and Jones AIAA 2015-1557

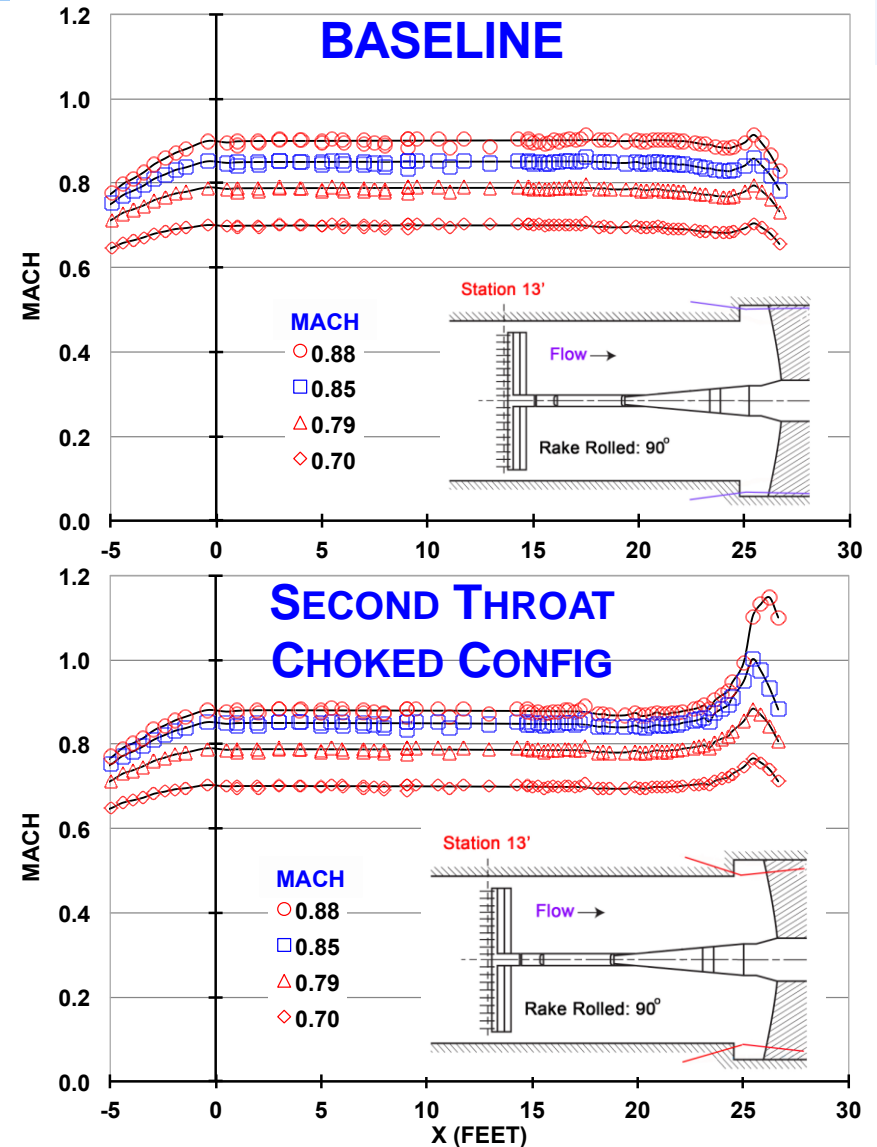
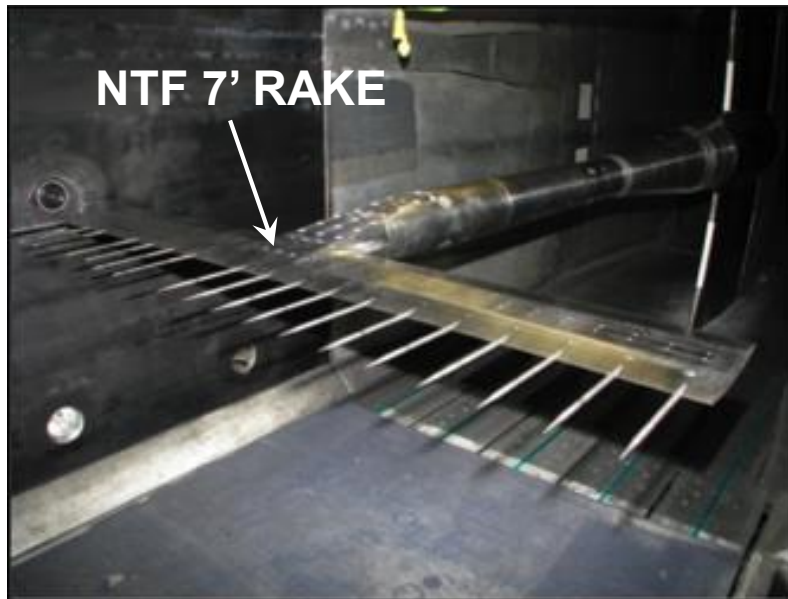
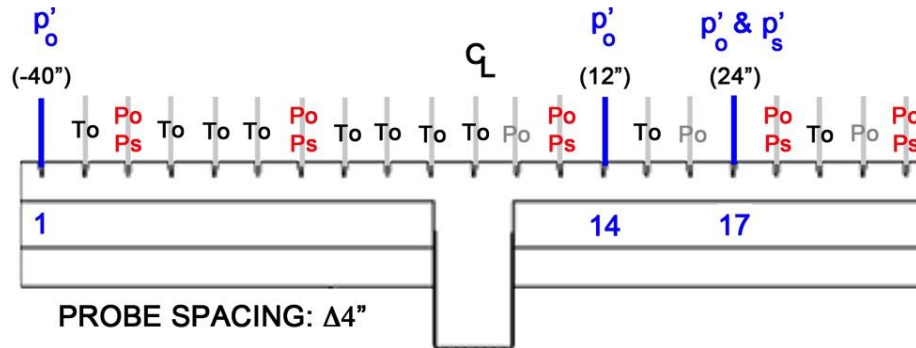
2nd THROAT W/ CENTERLINE PIPE CONFIGURATION



RAKE TEST



RAKE DATA TO BE USED FOR EVALUATING FLOW UNIFORMITY

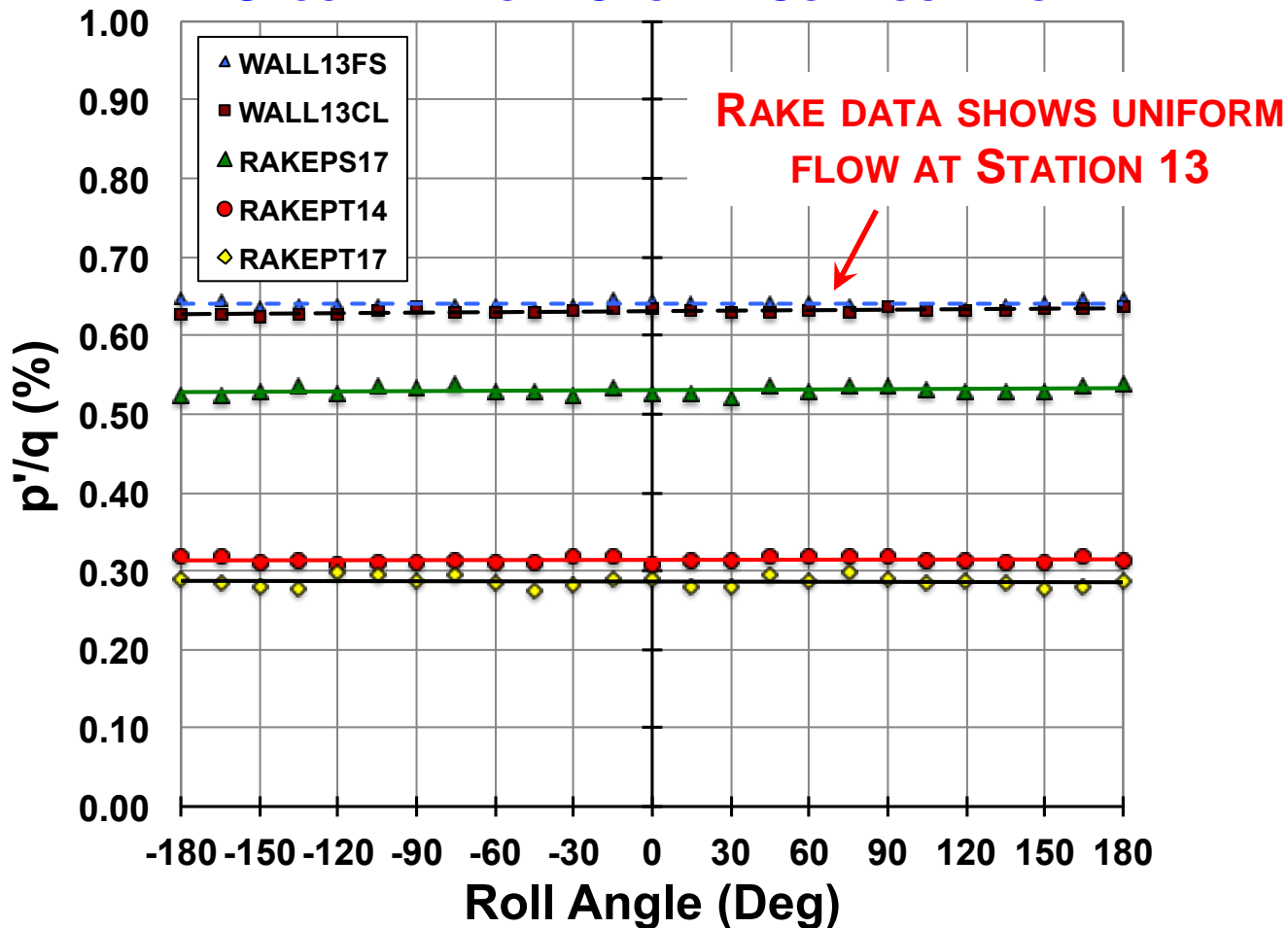


RAKE TEST SECTION UNIFORMITY



$M = 0.85$, $Re/ft = 49.4 \times 10^6$, $T_o = -251^\circ F$, $P_o = 44$ psia

SECOND THROAT CHOKED CONFIGURATION

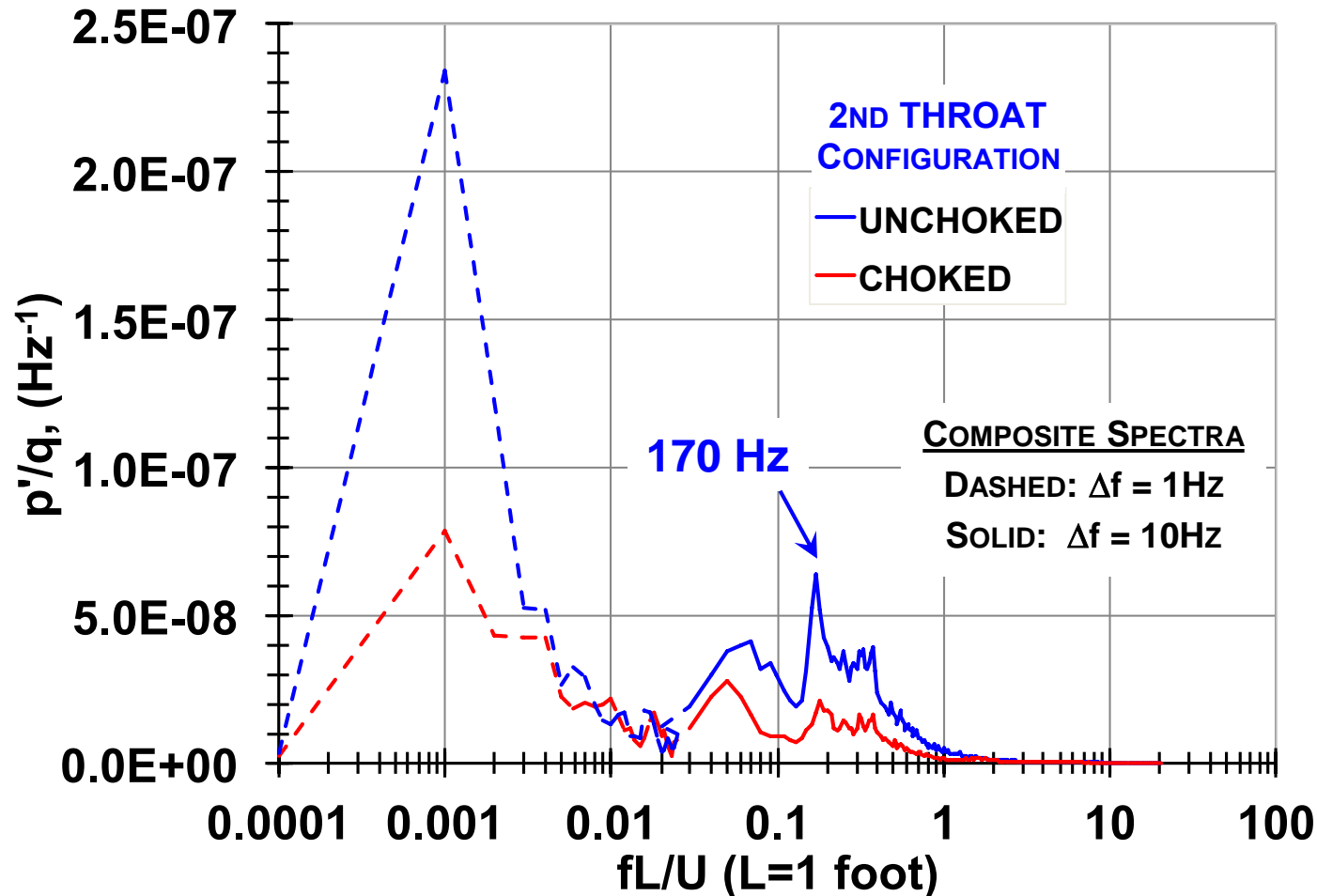


COMPARISON OF CHOKED AND UNCHOKED CONDITIONS

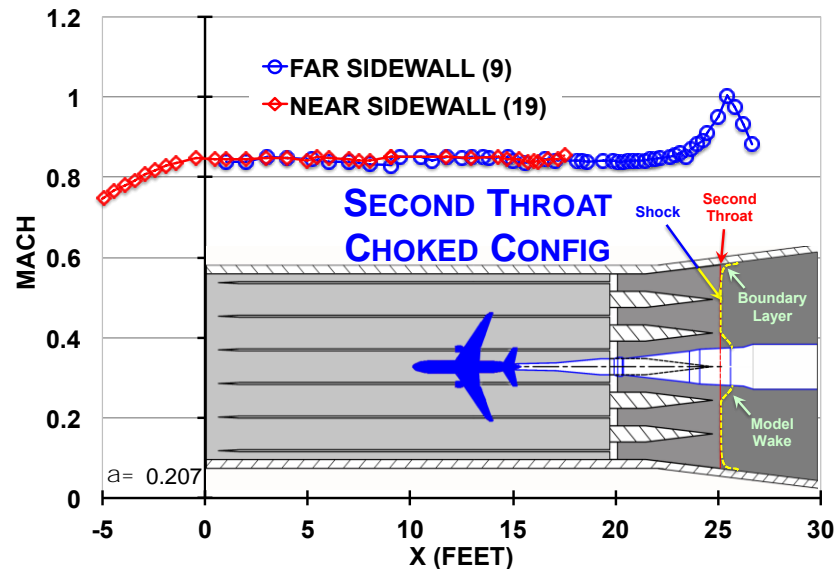
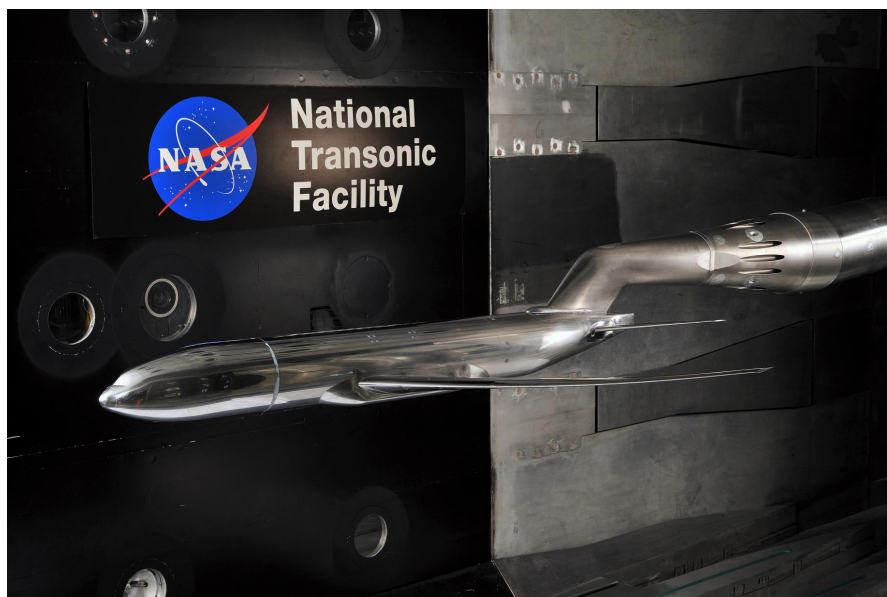
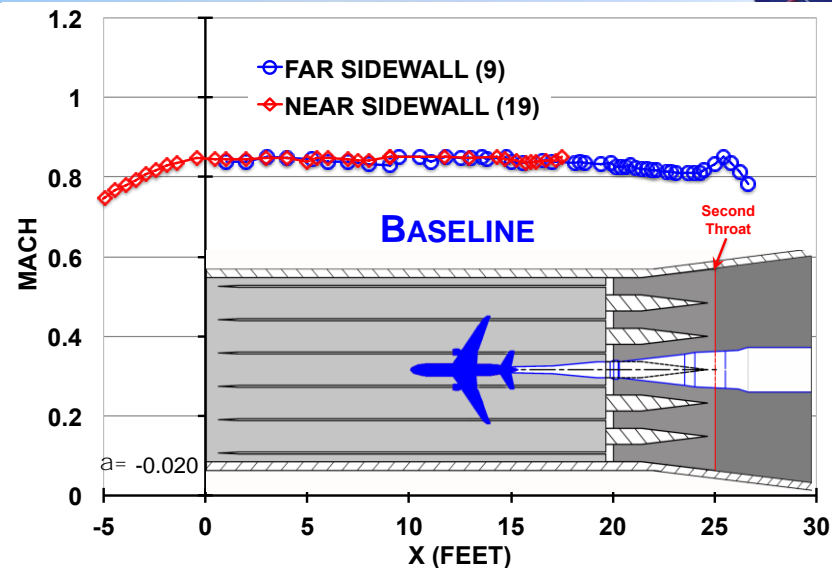
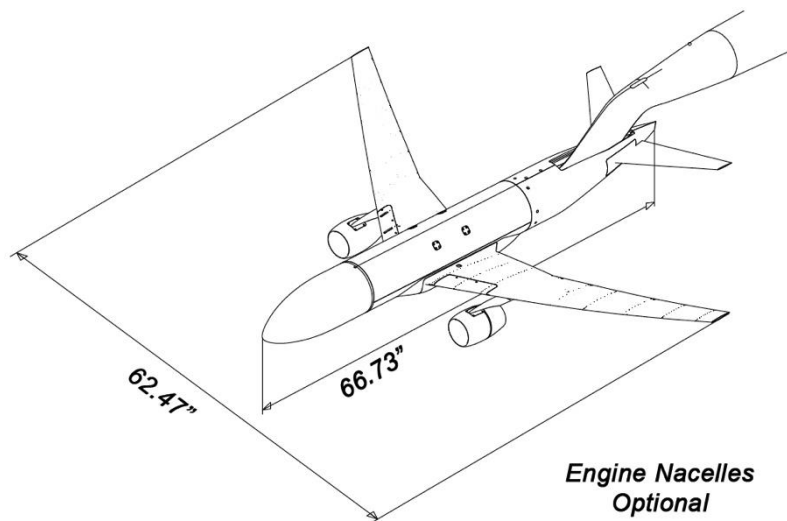


RAKE p'_{17} Static DATA STATION 13'

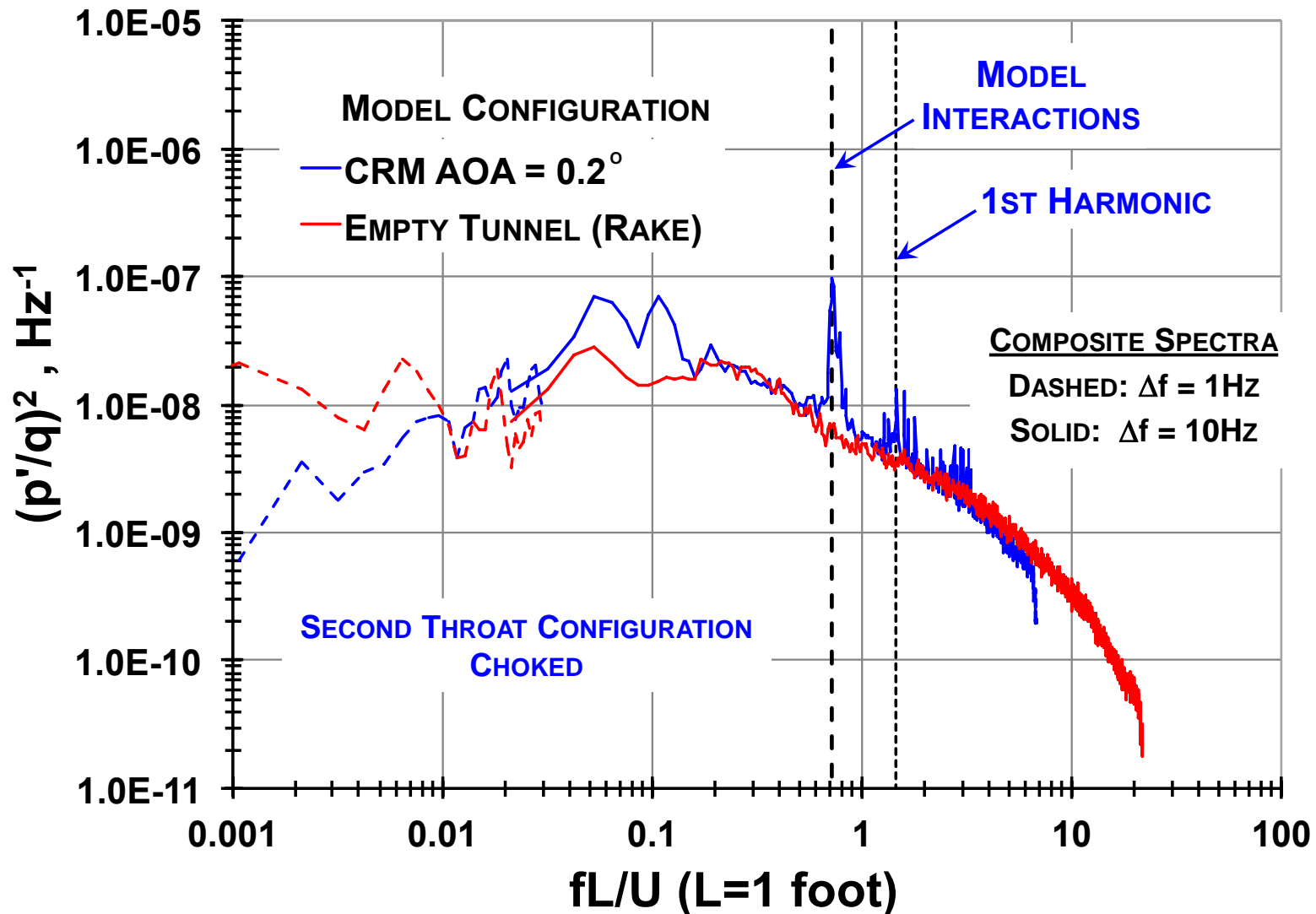
$M = 0.85$, $T_o = 120^\circ\text{F}$, $P_o = 30$ psia, $q = 9.5$ psi, Roll angle $= 0^\circ$



COMMON RESEARCH MODEL TEST



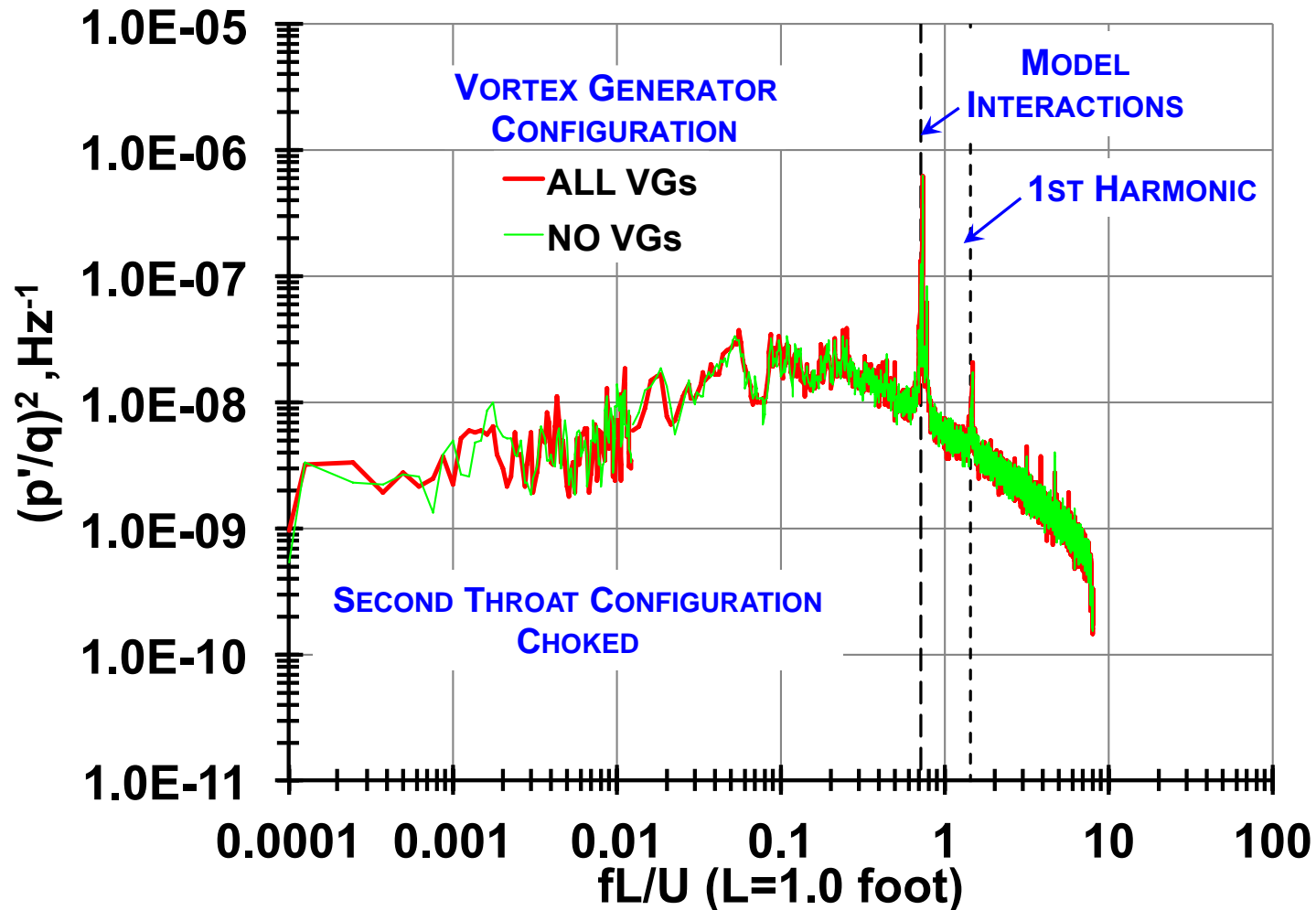
INFLUENCE OF COMMON RESEARCH MODEL



VGs INEFFECTIVE WITH COMMON RESEARCH MODEL

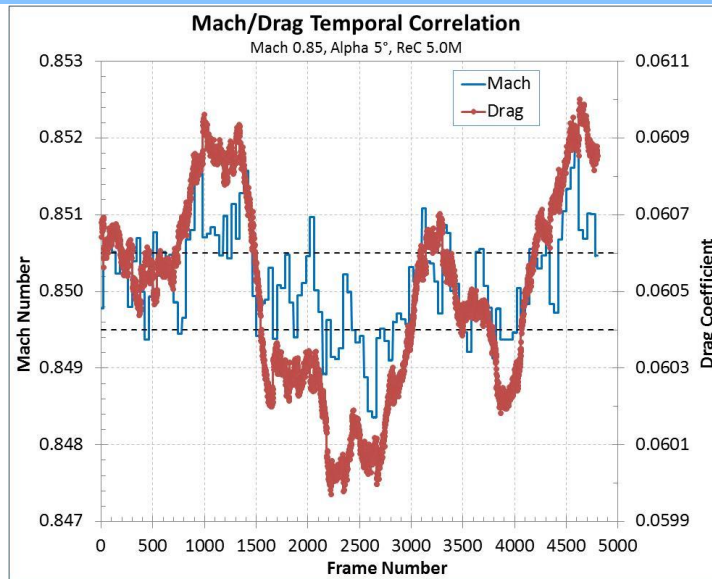


Mach = 0.85, $Re_c = 10.0 \times 10^6$, $q = 12$ psi, $T_o = -50^\circ\text{F}$, $P_o = 38$ psi



NTF TEST 218 – CRM

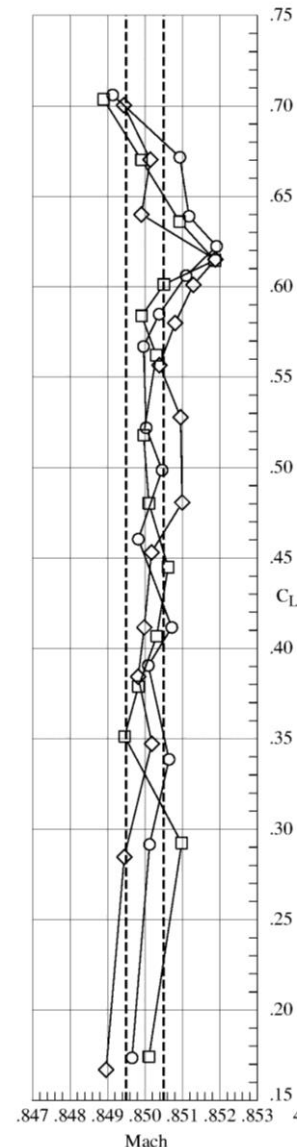
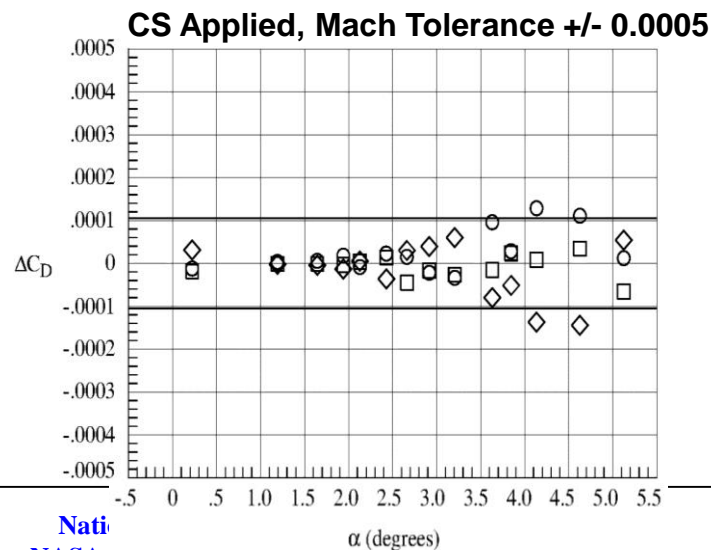
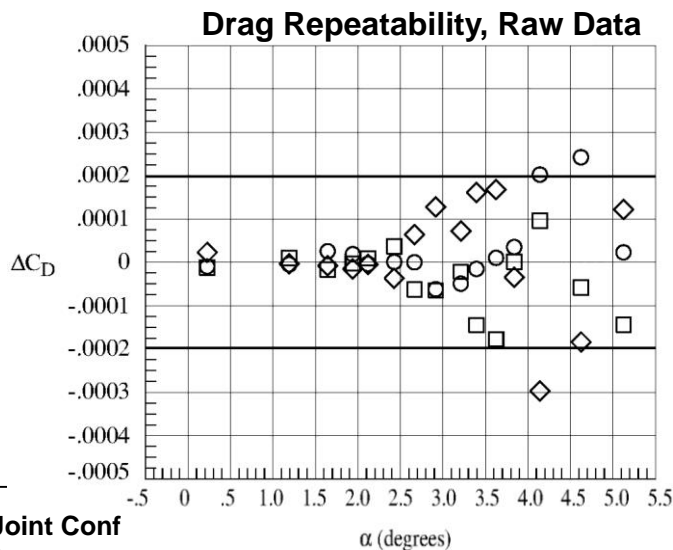
CONDITIONAL SAMPLING METHODOLOGY AND EFFECT



Measured drag force varies directly with measured Mach number variation

Mach variation increases at high AoA, probably due to separation (plot shows 3 repeat runs)

Conditional sampling only accepts and averages frames within a given Mach tolerance – given the correlation between Mach and drag, repeatability is improved



SUMMARY



- **STARBUKS effort finished testing 27 September**
 - Check Standard Model (CSM), Common Research Model (CRM), Centerline Pipe Calibration (CPC) and Flow Uniformity tests (Rake) have been completed
- **FIDO effort continues**
 - Desired data quality levels were achieved within series using computational and/or physical means
 - Preliminary Mach number variability to less than ± 0.0005
 - Preliminary C_D repeatability to less than ± 0.0001 at cruise conditions
- **Physical Means include VGs and 2nd Throat**
 - VGs were very effective in stabilizing low frequency characteristics associated with the high speed diffuser for empty tunnel configurations
 - VGs were NOT effective for the CRM transonic configuration
 - Choked second throat stabilized low frequency wind tunnel characteristics improving Mach number stability

QUESTIONS?



RESEARCH TEAM

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Scott Goodliff

Matthew Bailey

Roman Paryz

AIAA 2015-1557 JONES, ET.AL.

AIAA 2015-0622 CHAN, ET.AL.

AIAA 2014-1481 Paryz

BACKUP SLIDES



National Transonic Facility
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FIDO PROJECTS AND TESTS



- **5 Major Projects**

- **Test Section Movables (2nd Throat)**
 - Tunnel configuration selection
 - Mach control methodology
 - 2nd throat actuation
- **Conditional Sampling**
 - Off-line [Complete]
 - On-line real-time
- **Increasing Access Housing Heating**
- **Proportional Liquid Nitrogen (LN2) Injection**
 - Optimized nitrogen injection
 - Minimize nitrogen system hammering
- **LN2 Pump Health Monitoring**

- **5 Experimental Entries**

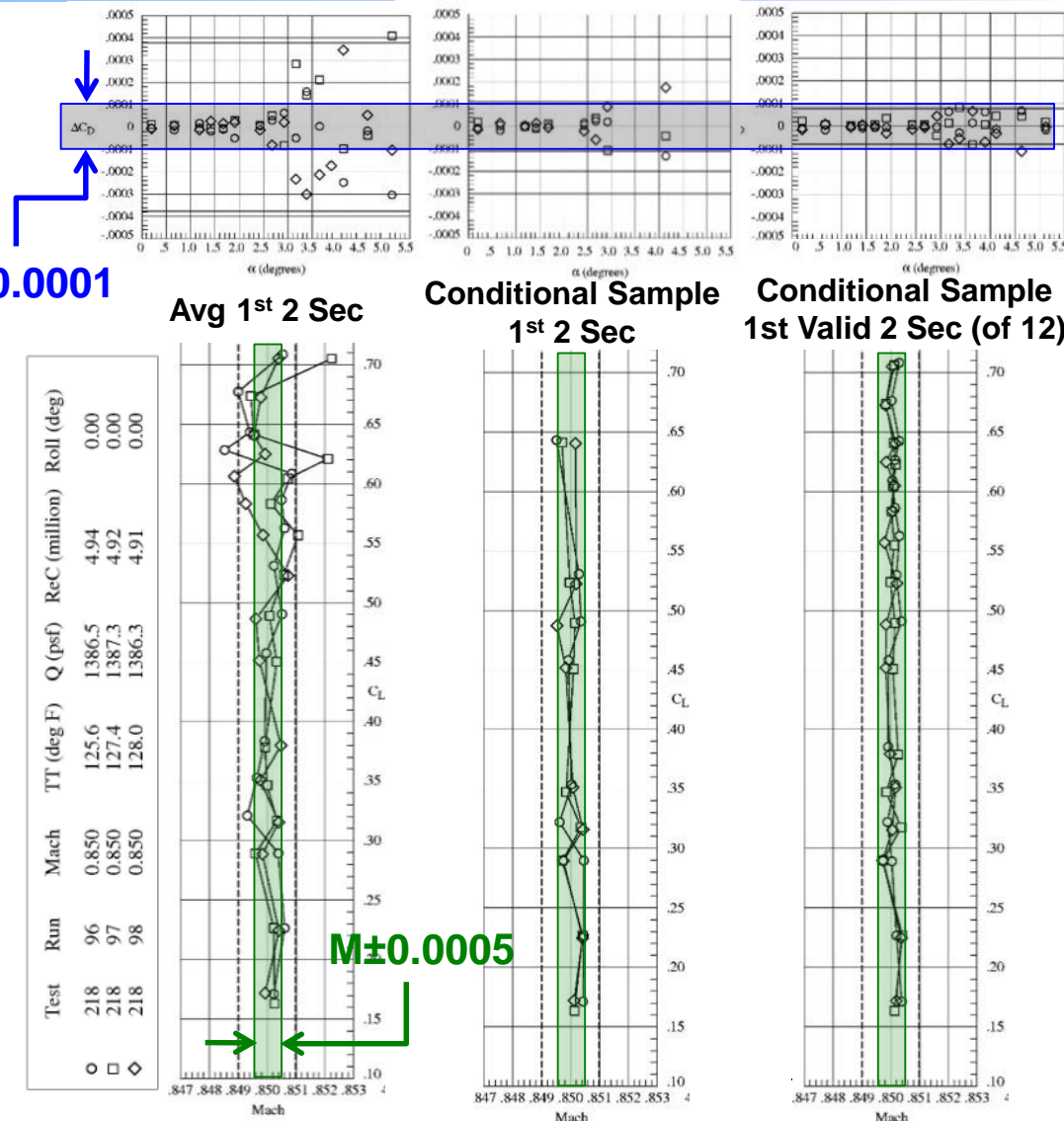
- **Test 219 Check standard [Pathfinder]**
 - Mach control methodology
 - Continuous sweep optimization
- **Test 216A&B Flow survey rake**
 - Validate RTD array
 - Verify turbulence reduction from STARBUKS [Deferred due to budget]
- **Test 220 Calibration extension**
 - Mach control methodology
- **Test 221 CRM validation**
 - Validation of combined system upgrades



CONDITIONAL SAMPLING



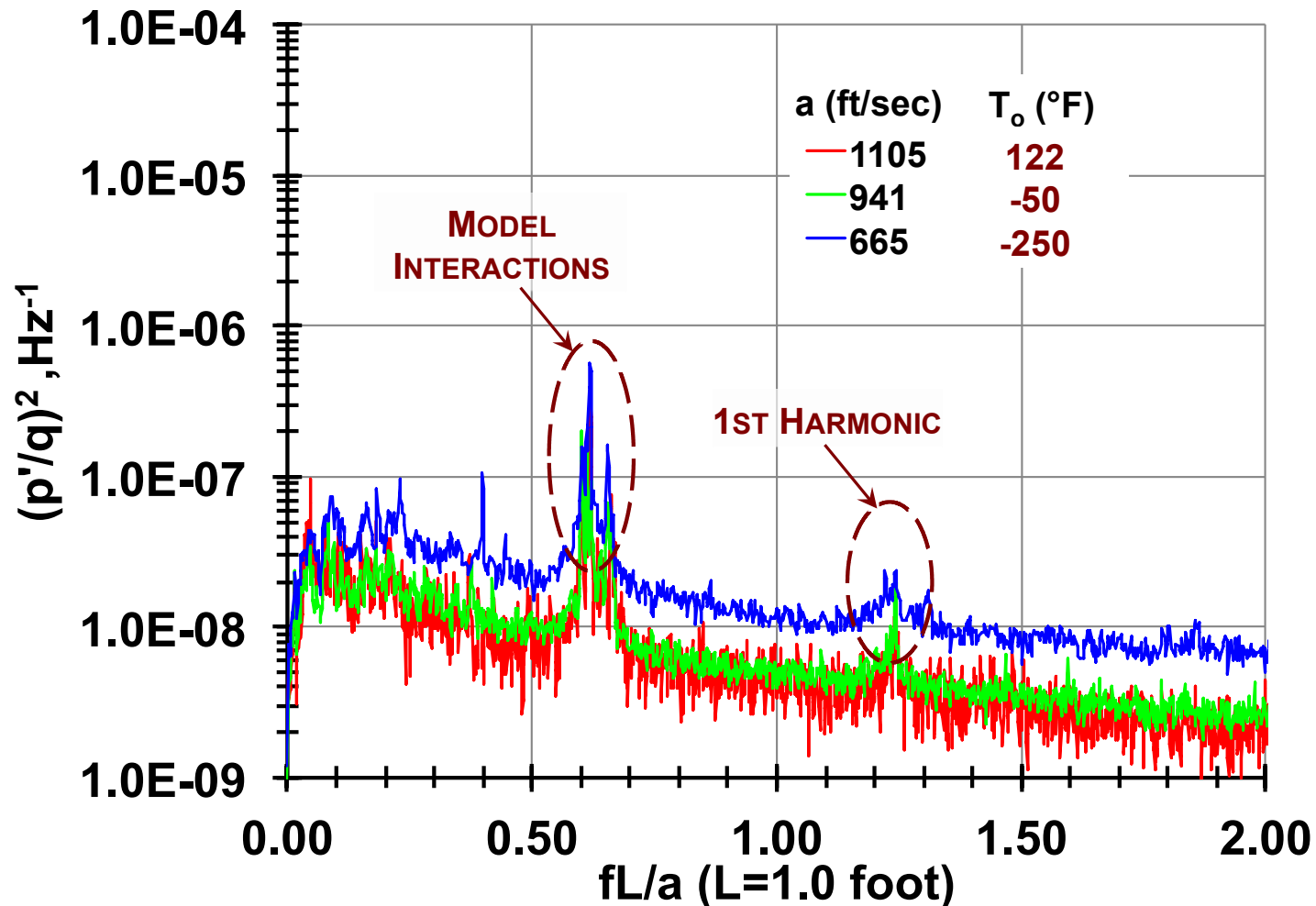
- **Improve data quality**
 - Reject data samples that do not meet requirements
- **Off-line: available**
 - Performance penalty due to longer data samples required
 - Need ~2 seconds of valid data
 - May need to acquire 10-12 sec
- **On-line: in development**
 - Stop acquiring data when samples meet specified criteria
 - Alleviates most of performance penalty



CORRELATION TO THE SPEED OF SOUND



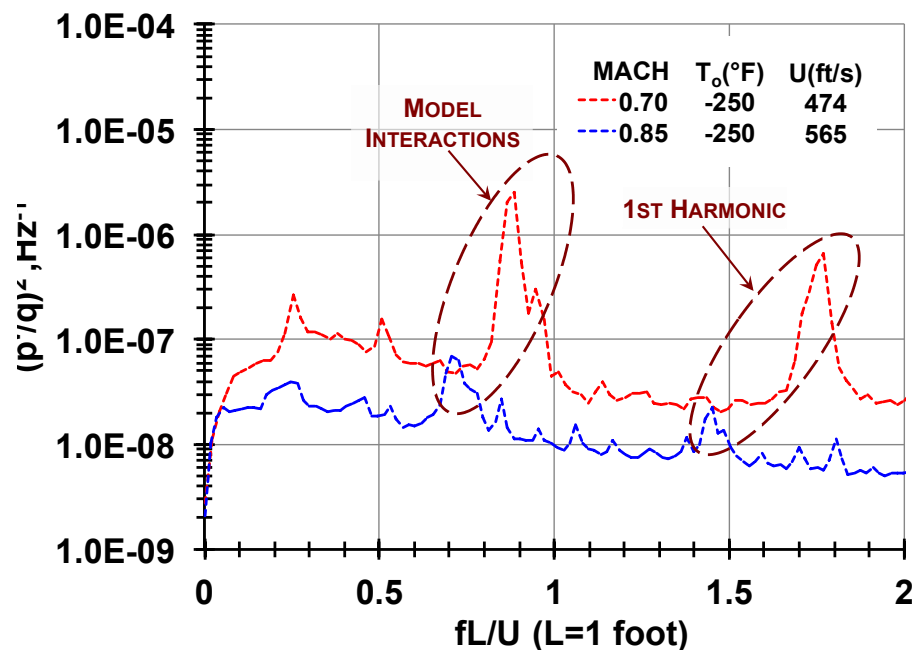
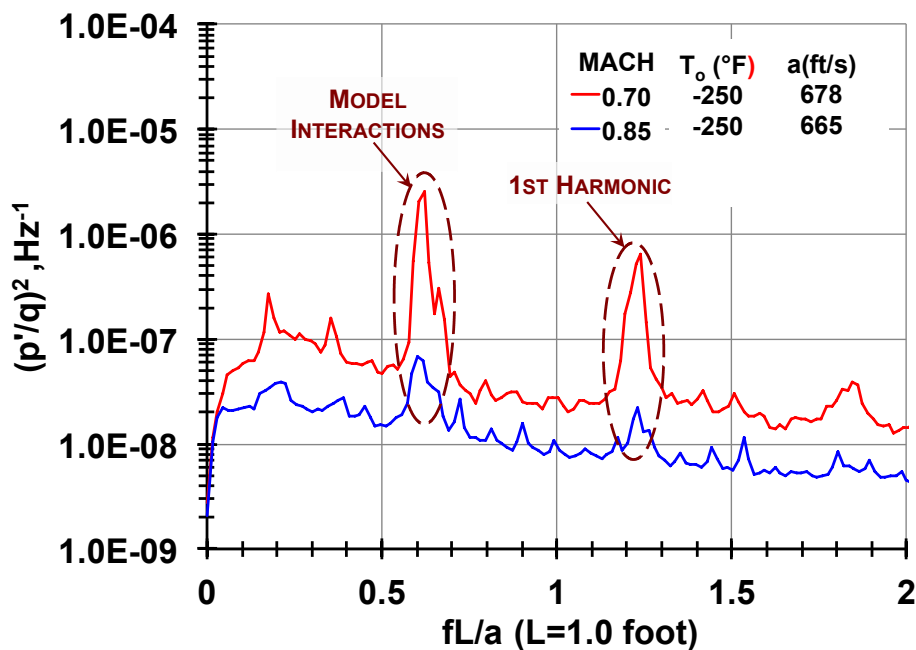
MACH = 0.85, Re_C = VARY, STATION 13 FAR WALL, CHOKED CONFIGURATION



CORRELATION WITH FREE STREAM VELOCITY



MODEL INTERACTIONS ARE NOT CORRELATED WITH MACH NUMBER



Station 13 Far Test Section Wall